

3.4.3 Flight Data Processing

HWCI 3.4.3.1 Flight Computer

- Requirements Traceability
 - DTRD Sections 3.3, 3.4
- Trade Studies
 - Selection of Processor Architecture
 - Rad Hard Full Function Space Rated Processors
 - Bus Architecture (Form Factor)
 - PCI, VME, STD32, PC104
 - Embedded Industrial Grade Flight Computers
 - Selection of PC104 Industrial Grade Supplier
 - Many COTS suppliers available
 - Ampro meets functional and performance requirements



3.4.3 Flight Data Processing

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HWCI 3.4.3.1 Flight Computer

- Functional and Performance Requirements
 - 1.0 PC/AT Functionality
 - a) Processor to be 486 class or above
 - b) COTS, Industrial Grade, Modular in Design
 - c) Meet power performance needs conservatively both in cost and CPU processing power
 - d) Processor to be upgradeable at low cost and with a minimal hardware redesign
 - e) Interface easily with other bus structures such as 1553 Dual redundant data bus, RS232, 422 etc.
 - f) Processor Board to take advantage of new technologies such as Flash Memory, Disk on chip Storage and SSD Storage



3.4.3 Flight Data Processing

HWCI 3.4.3.1 Flight Computer

- Functional and Performance Requirements (cont.)
 - 2.0 COTS Operating System Software Friendly
 - a) VX Works, Lynx, Qnx, etc.
 - b) Bios to be Programmable
 - 3.0 Environment Operational Requirement
 - a) 100,000 130,000ft altitude (mid-latitude)
 - b) -45c to +85c Temperature Range



3.4.3 Flight Data Processing

HWCI 3.4.3.1 Flight Computer

HWCI Description

- Ampro Core Module /4dxi 133mhz 486DX4 Processor
- PC/104 Compliant form factor
- Size: 3.6 x 3.8 x 0.9 in. Weight: 3.4 oz.
- + 5v + -5% 5 watt max
- Full function PC/AT with up to 52meg onboard Dram
- Ruggedized BIOS, Bootable solid State Disk on Chip
- Onboard Flash to 1meg byte
- Batteryless boot, watchdog timer, Embedded BIOS
- Advanced Power Management, +5v operation
- Extended Operating Temp Range -40c to +85c



3.4.3 Flight Data Processing

HWCI 3.4.3.1 Flight Computer

- Risk Assessment & Mitigation/Reliability
 - This core module has flown on F-16, 737, 757 research aircraft and on sounding rockets as well
 - Fourteen Year History, Highly Reliable, Proven embedded industrial grade PC/AT System
 - COTS Product



3.4.3 Flight Data Processing

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HWCI 3.4.3.2 PC104 Power Supply

- Requirements Traceability
 - DTRD Sections 3.3, 3.4
- Trade Studies
 - Build in-house
 - Select from multiple COTS vendors
- Key Functional and Performance Requirements
 - PC/104 form factor
 - COTS available and operate in embedded industrial applications
 - Extended Temperature operation -40c to +85c
 - Efficiency up to 95%
 - Output power of 10 amps at 5 volts



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HWCI 3.4.3.2 PC104 Power Supply

- HWCI Description
 - Diamond PC/104 Power Supply
 - HE104-512 WITH OPT-512 & TAC HIGH POWER
 - Clean & filtered Power for PC/104 bus
 - "Load Dump" Transient protection
 - High power converter, high efficiency to 95%
 - + 5v, +/- 12v, 50 watts
 - Operating temp. to -60c to +85c under ideal conditions
 - Fully PC/104 compliant
 - AC Termination instead of DC termination
- Risk Assessment & Mitigation/Reliability
 - COTS Product
 - Flown on NASA Delta Clipper and survived the Crash
 - Redundant items flown



3.4.3 **Flight Data Processing**

HWCI 3.4.3.3 PC/104 Enclosure

- Requirements Traceability
 - DTRD Section 3.4
- **Trade Studies**
 - Build in-house
 - Select from multiple COTS suppliers
- Functional and Performance Requirements
 - Shall have easy access to cards
 - Shall be suitable for heat sinking
 - Shall support rail mounting of cards
 - Shall provide EMI isolation
 - Maintain functionality in ULDB environment for mission duration



3.4.3 Flight Data Processing

HWCI 3.4.3.3 PC/104 Enclosure

- HWCI Description
 - Parvus Corp. Chassis Cage System
 - 6" x 7" x10" Long Extrusion, EMI gaskets
 - 8" Aluminum Rails System to hold up to 11 cards
- Risk Assessment & Mitigation/Reliability
 - COTS Product
 - Redundant items flown



3.4.3 Flight Data Processing

HWCI 3.4.3.4 1553 Interface Card

- Requirements Tractability
 - DTRD Sections 3.4.1.1, 3.4.4.1, 3.5.1.1
- Trade Studies
 - Build in-house
 - Select from multiple COTS suppliers
- Functional and Performance Requirements
 - Shall interface with science Computer for data transfer
 - Shall interface with TDRSS Transmitter / Receiver
 - Meet MIL-SPEC Standard & industrial Grade Environmental Specifications



3.4.3 Flight Data Processing

HWCI 3.4.3.4 1553 Interface Card

- Functional and Performance Requirements (cont.)
 - Dual redundant bus for Failure backup of the Flight computers and data bus
 - PC/104 Form Factor
 - Emulation Capabilities
 - Configurable to Bus controller, Bus Monitor, and remote
 Terminal Functionality via remote software download
 - Software (RTOS ready, Drivers written and Proven)
 - Input power shall be 4 watts maximum
 - Maintain functionality in ULDB environment for mission duration



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HWCI 3.4.3.4 1553 Interface Card

- HWCI Description
 - SBS Avionics Technologies
 - ABI-PC-104xt 1553 interface Card
 - MIL-STD 1553 Full function programmable in three operational modes (BC,BM,RT) Remotely
 - Xlink Gate array for protocol processing
 - Map and Sequential Monitor modes@ 1 Sec time sampling, double buffers for Rt. data logging etc.
 - Digital signal processor routing 1553 words, and packing/decoding messages to program firmware.
 - (RTOS Friendly), VxWorks ready, Full Library of Software Drivers already written and in use
 - Onsite Technical support available



3.4.3 Flight Data Processing

HWCI 3.4.3.4 1553 Interface Card

- Risk Assessment & Mitigation/Reliability
 - These Cards have a long proven history in Military Avionics Projects such as F-22,F16, B2 and in NASA International Space Station Freedom and in flight ground station at GSFC
 - COTS Product
 - Redundant items flown



Processing

3.4.3 **Flight** Data

HWCI 3.4.3.5 RS-232 Interface Card

- Requirements Traceability
 - DTRD Section 3.4
- **Trade Studies**
 - Build in-house
 - Select from multiple COTS suppliers
- Functional and Performance Requirements
 - PC/104 Form factor
 - Multiple Serial Ports per Card (need 10 to 12 total)
 - Input power shall be 0.4 watts maximum
 - RTOS Friendly Software drivers written and proven
 - Shall support interrupt sharing



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HWCI 3.4.3.5 RS-232 Interface Card

- HWCI Description
 - Diamond Systems Corp.
 - Emerald-MM v3 Quad 232/422 interface Card (stackable)
 - 4 ports per board, PC/104 form-factor
 - Dual 20 pin headers (2 ports per)
 - Flexible Address and interrupt selection
 - Interrupt sharing
 - +5volt operation, low power
 - 16c554 quad serial port IC that supports data rates to
 115kbps and contains 16-byte programmable FIFO buffer to minimize processor overhead
 - Short circuit protection (Indefinite, all outputs)
 - Four layer circuit board (split power and grounds) for noise reduction



3.4.3 Flight Data Processing

HWCI 3.4.3.5 RS-232 Interface Card

- Risk Assessment & Mitigation/Reliability
 - Has Flown on the Space Shuttle used in three separate experiments
 - One Experiment was so successful and reliable that it is scheduled to be one of the first payloads on the Space station
 - COTS Product
 - Redundant items flown



3.4.3 Flight Data Processing

HWCI 3.4.3.6 SCSI Interface Card

- Requirements Traceability
 - DTRD Sections 3.4.2
- Trade Studies
 - Build in-house
 - Select from multiple COTS suppliers
- Functional and Performance Requirements
 - PC/104 form factor
 - Shall interface to on-board hard disk drives
 - Shall provide two serial ports
 - Input power shall be 0.7 watts maximum



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HWCI 3.4.3.6 SCSI Interface Card

- HWCI Description
 - Ampro MM2/SES
 - Dual serial controller: 16550 compatible, FIFO buffered, one RS232C and one RS232C/485 channel
 - SCSI-II interface: Up to 10 MBytes/sec data based on popular Adaptec AIC6370 controller
 - 130 mA at 5V
 - -40 to +85 degrees C operating temperature
- Risk Assessment & Mitigation/Reliability
 - COTS Product
 - Redundant items flown